

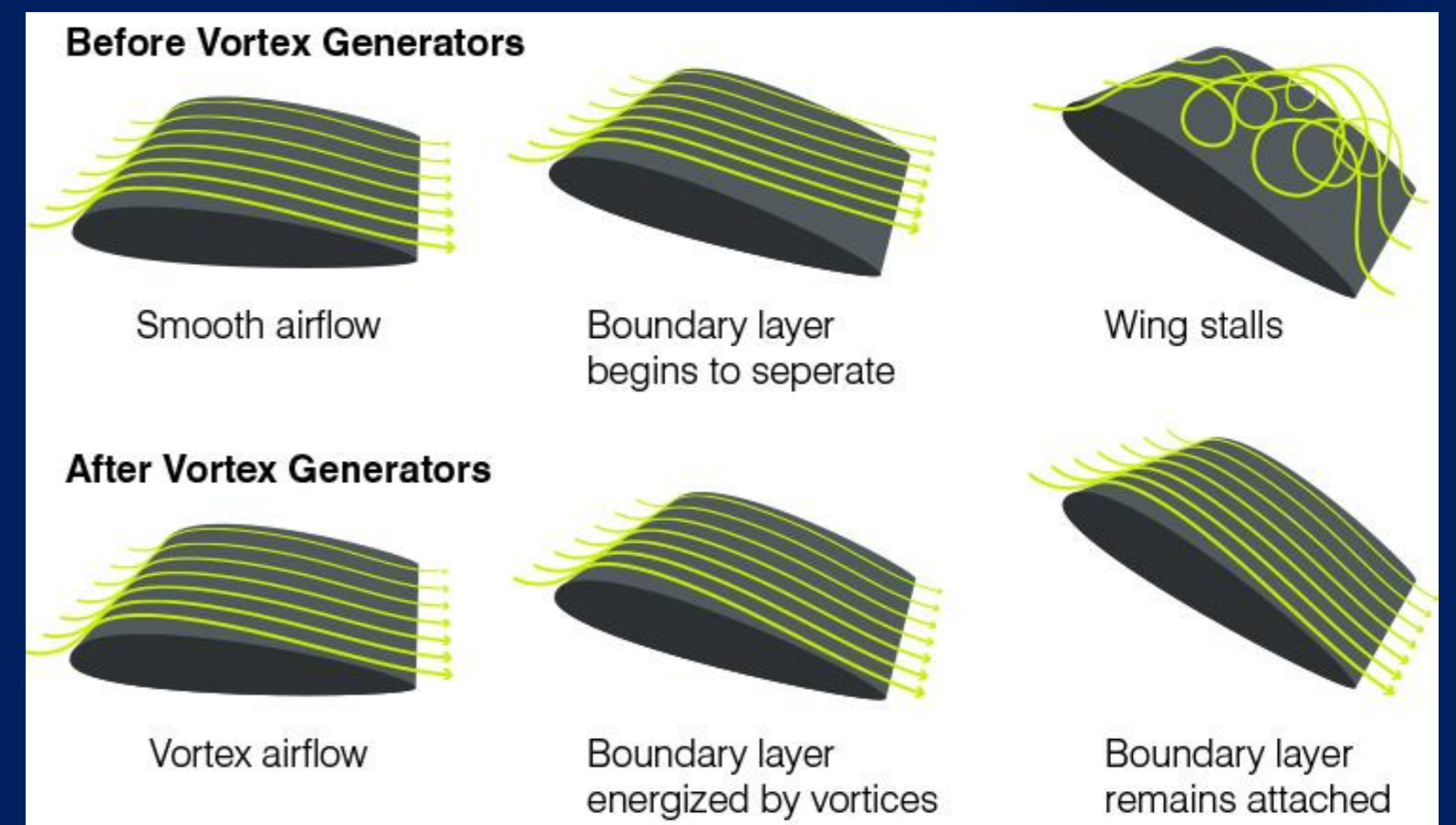
XFOIL for Vortex Generators: Fast aerodynamic analysis of Airfoils equipped with VGs

Abhratej Sahoo
FPT – Wind Energy
Prof. Dr. ir. Carlos Ferreira
Dr. Wei Yu
Dr. Gerard Schepers (TNO)
Dr. A. K. Ravishankara (TNO)
A.Sahoo@tudelft.nl



Background:

Vortex generators delay flow separation on thick root sections of ever-growing wind turbine blades. Turbine developers want to optimise VG arrays with fast and accurate tools. But, CFD simulations or wind tunnel measurements for a single configuration can take days and weeks.

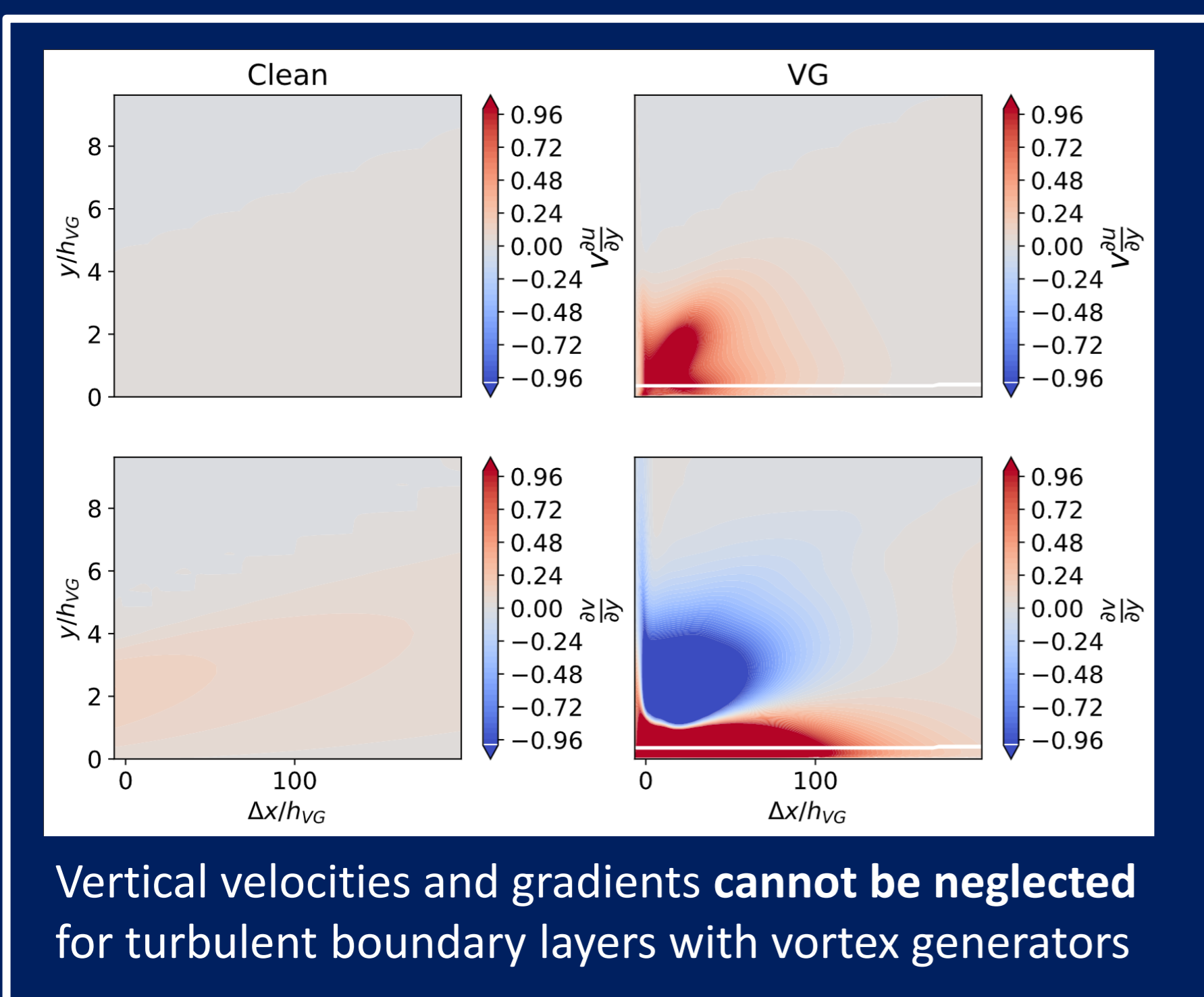


Goals of my PhD:

- To extend an integral boundary layer framework to vortices in boundary layers to create **XFOIL for vortex generators**.
- To design an innovative blade accounting for vortex generators

Steps:

- Computational Fluid Dynamics and Particle Image Velocimetry of vortices in flat plate boundary layers
- Modifying the integral boundary layer equations with additional momentum terms
- Establishing the relationship between VG geometry and inflow conditions to added momentum



$$\frac{d\theta}{d\xi} = \frac{C_f}{2} - (H + 2) \frac{\theta}{u_e} \frac{du_e}{d\xi} + \text{Added Momentum}$$

$$\frac{dH_k}{d\xi} = \frac{2C_D}{\theta} - \frac{H_k C_f}{\theta} - \left(\frac{2}{H_k} + 1 - H \right) \frac{H_k}{u_e} \frac{du_e}{d\xi} + \text{Added Energy}$$

$$\frac{\partial(u_e C_\tau)}{\partial \xi} = \frac{C_\tau u_e}{\delta} K_c \left(\left(C_{\tau EQ} \frac{1}{2} \right) - C_\tau \frac{1}{2} \right) - C_\tau \frac{\partial u_e}{\partial \xi} + \text{Added Turbulence}$$

These integral boundary layer equations, derived from Navier Stokes equations, **now need extra terms** for added momentum, energy, and turbulence

Prototype: A model for the shape factor (representing added momentum) produces accurate lift and drag for airfoils with vortex generators

