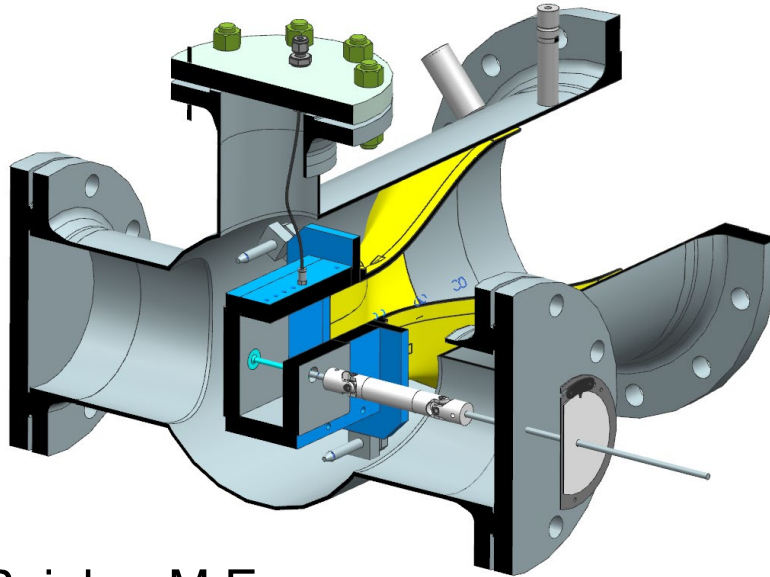




**NICFD 2020**  
for Propulsion & Power

# Performance of a Rotatable Cylinder Pitot Probe in High Subsonic Non-Ideal Gas Flows



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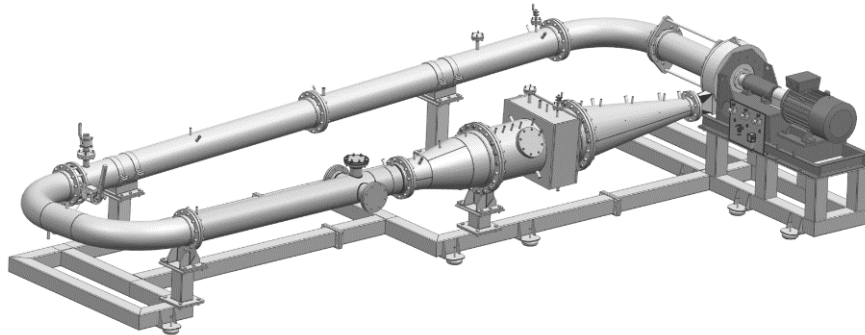
**FH MÜNSTER**  
University of Applied Sciences

**Paper ID: 8**

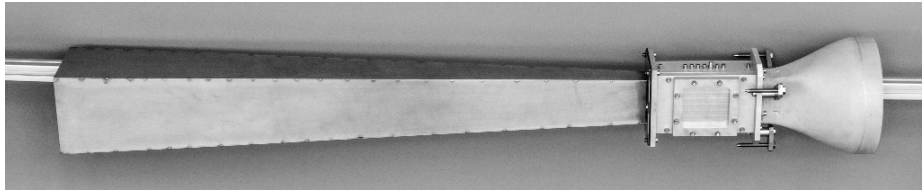
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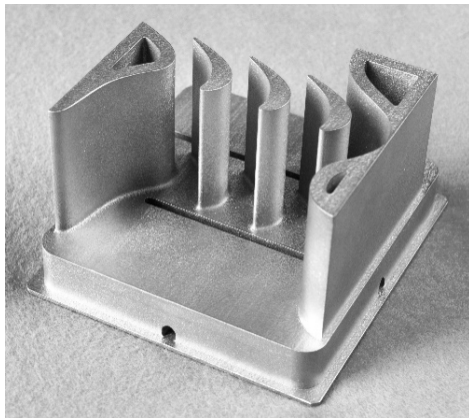
# Introduction & Motivation



Closed Loop Organic vapor  
Wind Tunnel (CLOWT)  
Fluid: Novec™ 649



Modular High Speed Test Section  
High Subsonic - Transonic



Printed turbine  
guide vanes

Linear blade cascade  
Mach number determination  
→ review of loss correlations  
→ affect of surface roughness

# Pitot Probes: Basics

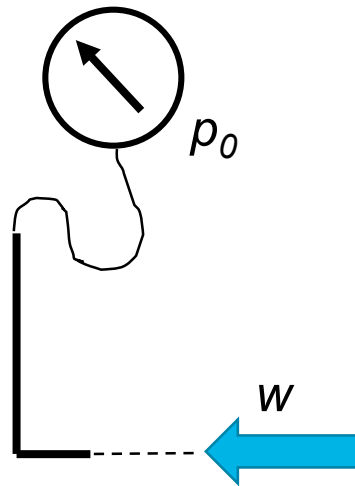
Definition: Device to measure stagnation (or total) pressure

Principle: Flow is brought isentropically to rest (stagnation)

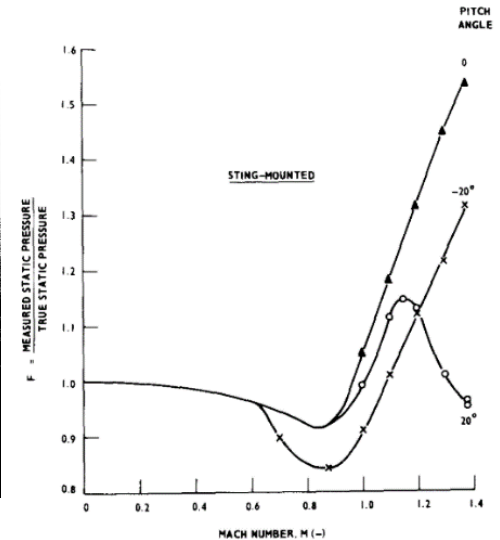
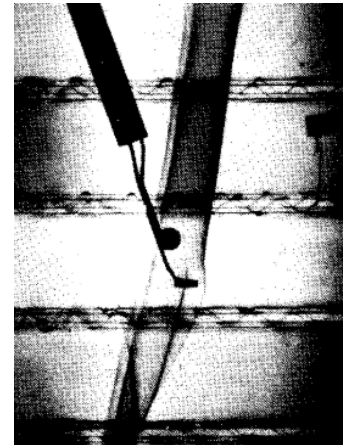
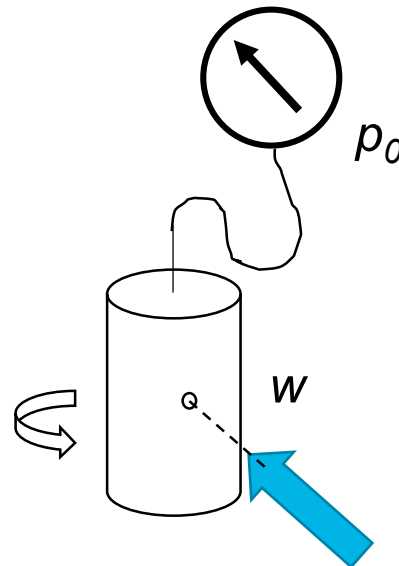
Purpose: Determination of flow velocity (+free static pressure)

Invention: Henri Pitot (1732)

## Classical L-Pitot Probe



## Cylinder Pitot Probe



Langford et al. 1982  
(issues for  $M > 0.4$  observed)

# Pitot Probes: Gasdynamics

Isentropic flow relations:

Perfect gas flows:

Subsonic:

$$M^2 = \frac{2}{\gamma-1} \left( \left( \frac{p_0}{p} \right)^{(\gamma-1)/\gamma} - 1 \right)$$

Supersonic:

$$\frac{p_0}{p} = \frac{\gamma+1}{2} M^2 \left( \frac{(\gamma+1)^2 M^2}{4\gamma M^2 - 2(\gamma-1)} \right)^{1/(\gamma-1)}$$

$p$ : free-stream static pressure

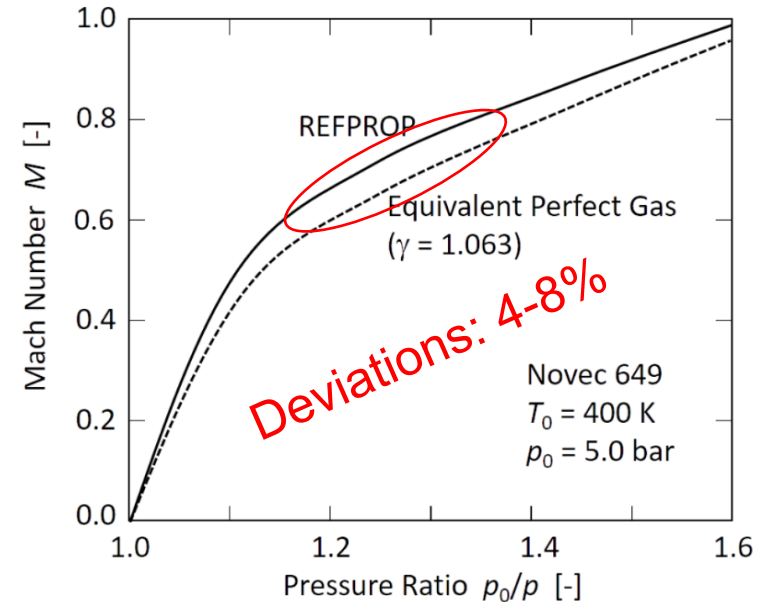
$p_0$ : Pitot probe total pressure  
(downstream of shock)

$\gamma$ : *Isentropic exponent*

Non-ideal gas flows:

- Suitable equations of state
- Solving mass & energy balance equations

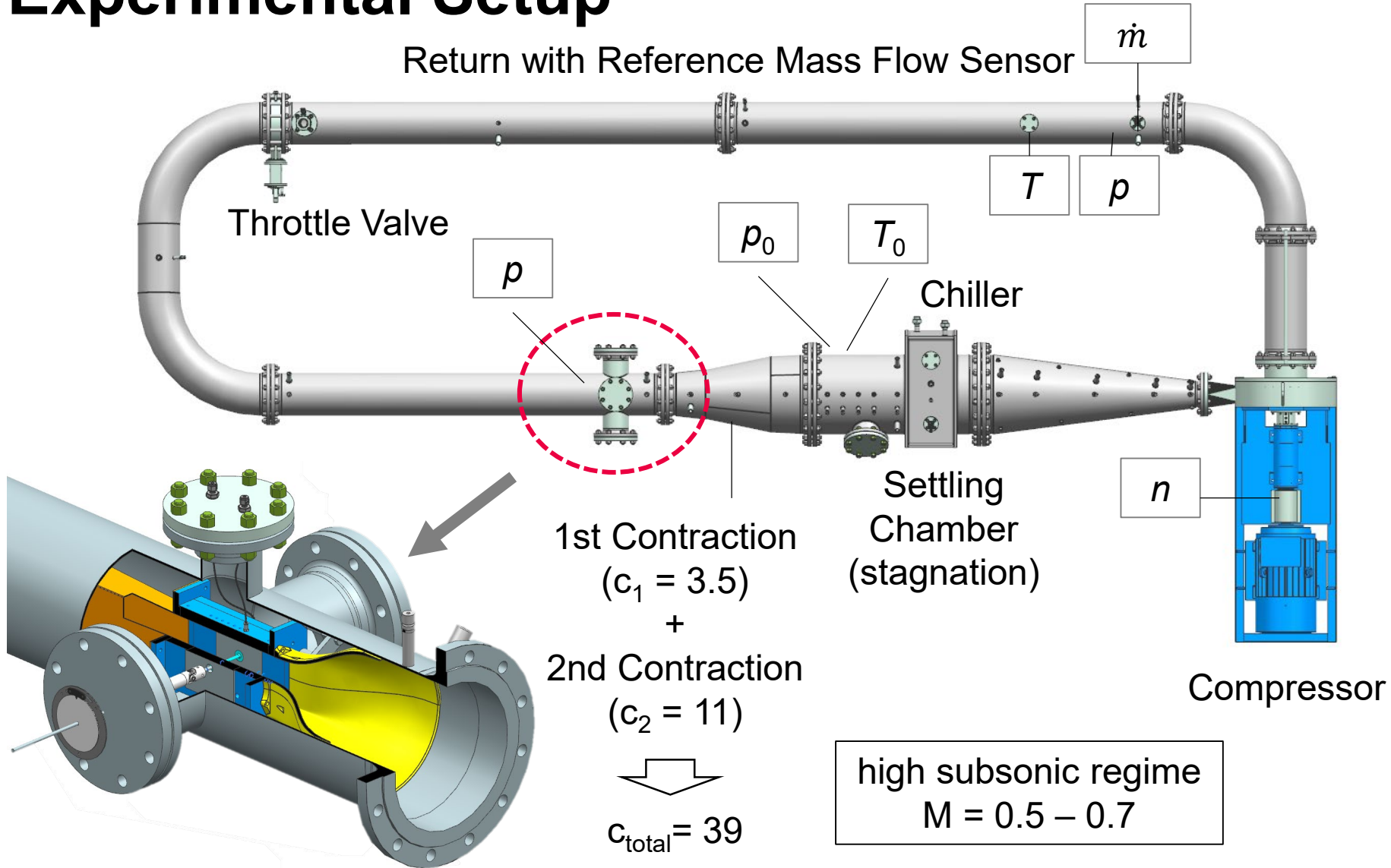
$$M = f(p, p_0, T_0)_s$$



*Method: Passmann M. et al. – NICFD 2016*

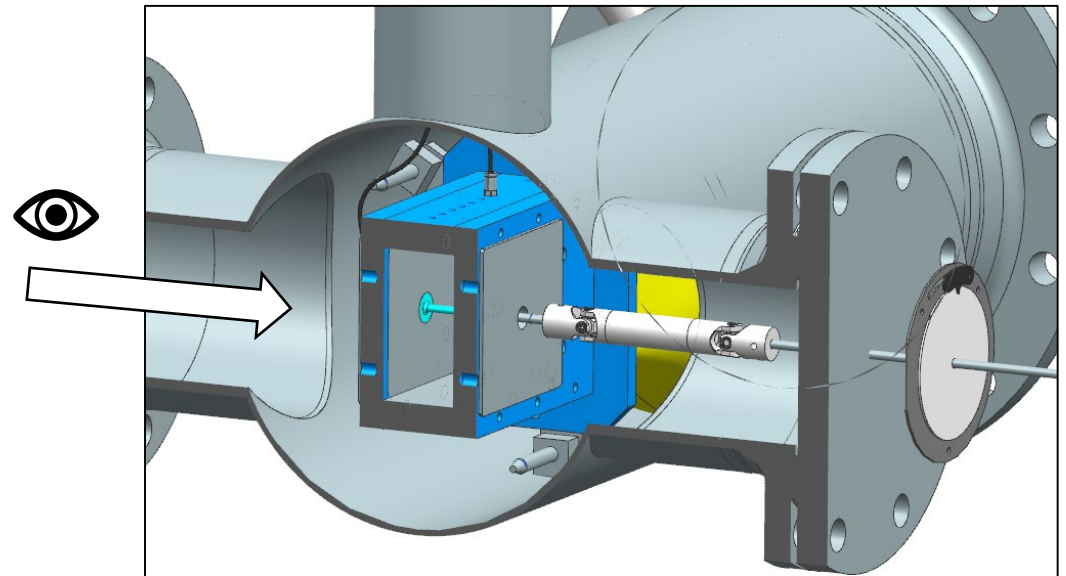
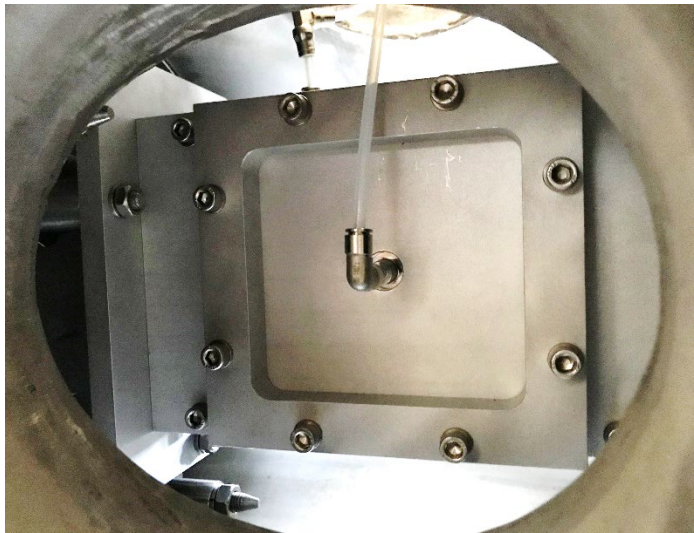
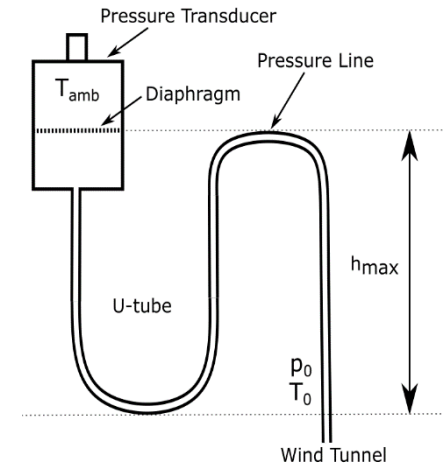
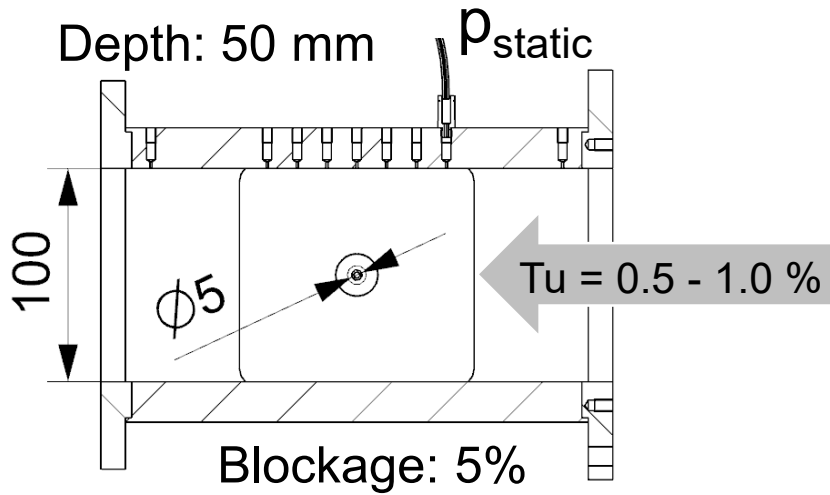
# Experimental Setup

Return with Reference Mass Flow Sensor





# Experimental Setup



# Experimental Setup

Cylinder Pitot Probe



Insert with flexible cardan shaft





# Experimental Setup

## Process parameters & fluctuations

Process parameter	Value
Pressure level $p$ [Mpa]	$0.294 \pm 0.017$
Temperature level $T$ [K]	$368 \pm 1.6$
Density $\rho$ [kg/m <sup>3</sup> ]	$34.5 \pm 2.2$
Compressibility factor $Z$	$0.88 \pm 0.01$
Speed of Sound $a$ [m/s]	$88.2 \pm 0.9$
Isentropic exponent $\kappa$ [-]	1.056
Dynamic viscosity $\mu$ [Pa·s]	$1.38 \cdot 10^{-5}$
Prandtl number $Pr$ [-]	0.77
Reynolds number $Re$ [-]	$1.8 \cdot 10^5 - 6.1 \cdot 10^5$
Mach number $M$ [-]	0.16 – 0.7

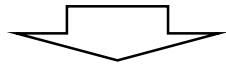
## Measurement uncertainties

Process variable	Uncertainty
Pressure $\Delta p/p$	<b>0.6 - 1.5%</b>
Temperature $\Delta T/T$	< 0.1%
Mach number $\Delta M/M$	<b>1.0 - 3.0%</b>
Reynolds number $\Delta Re/Re$	1.8 – 4.8%
Compressibility factor $\Delta Z/Z$	2.4%
Pressure Coefficient $\Delta C_p/C_p$	1.0%

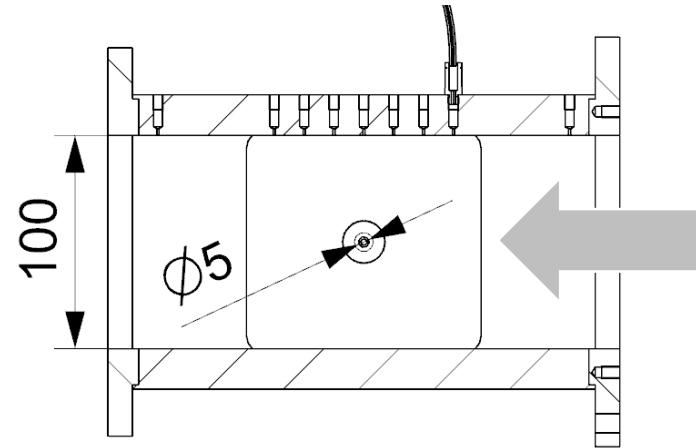
Fluid: Novec™ 649 by 3M™

# Blockage correction

Blockage factor  $d/H = 5 \%$



Perturbed velocity field affects:  
 $C_p$  and  $w$

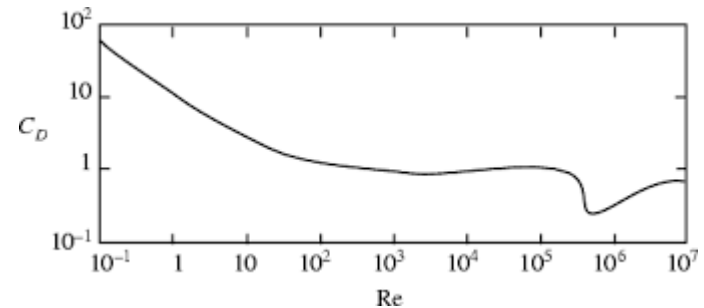


Blockage parameter  $\epsilon$  (Wyler 1975)

$$\epsilon = \frac{w - w_{Pitot}}{w_{Pitot}} = \frac{C_D}{2(1-M^2)} \frac{d}{H}$$

$$C_p - 1 = \left( \frac{w_{Pitot}}{w} \right)^2 (C_{p,Pitot} - 1)$$

$$C_p = \frac{p_{pitot} - p}{p_0 - p}$$

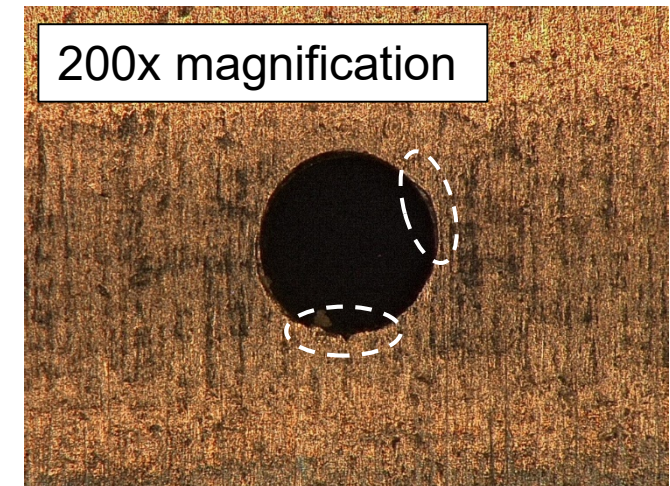
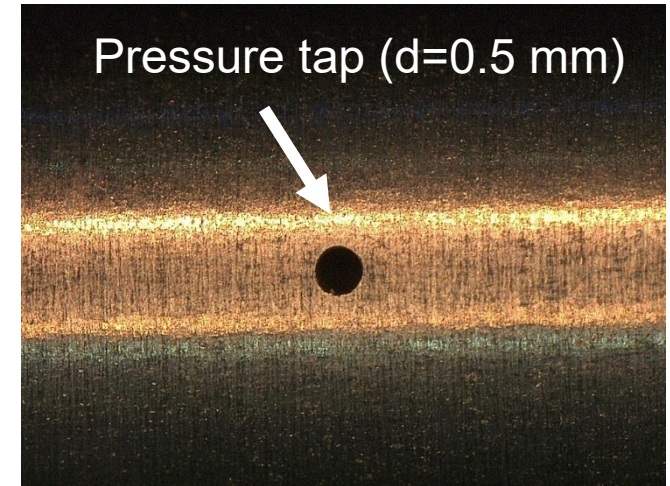
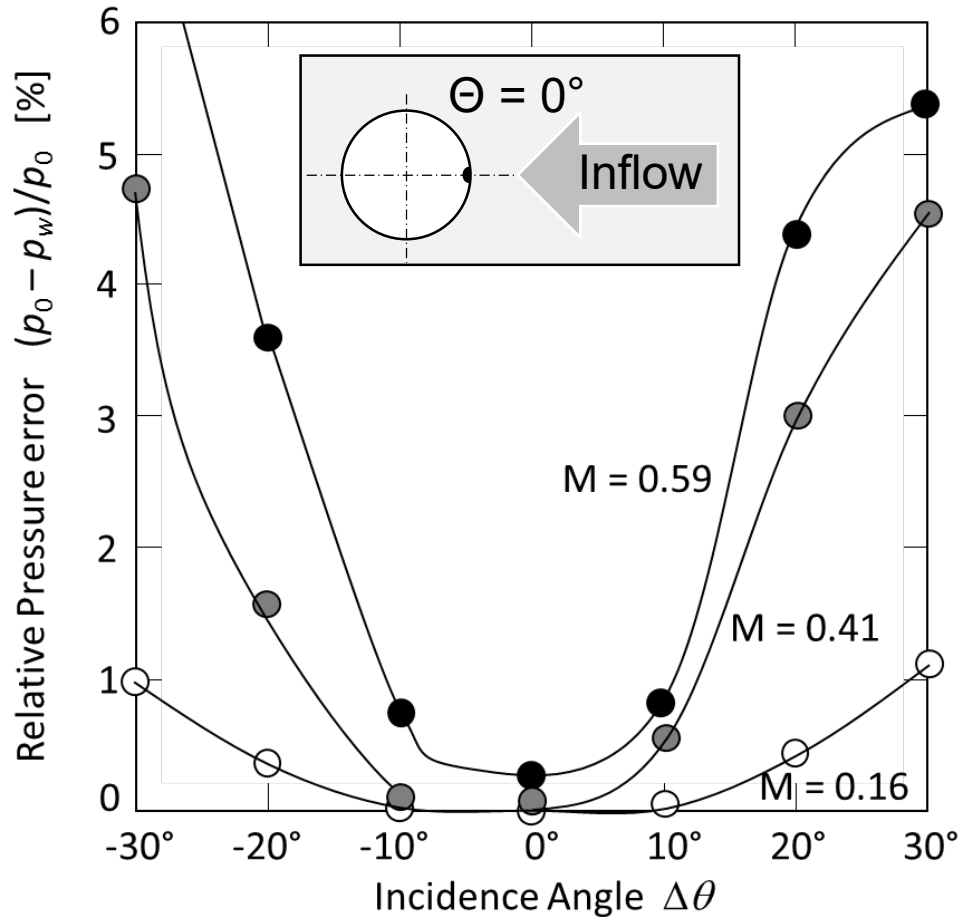


$$C_D = 1.15 + 0.75 (M - 0.2) \quad ?$$

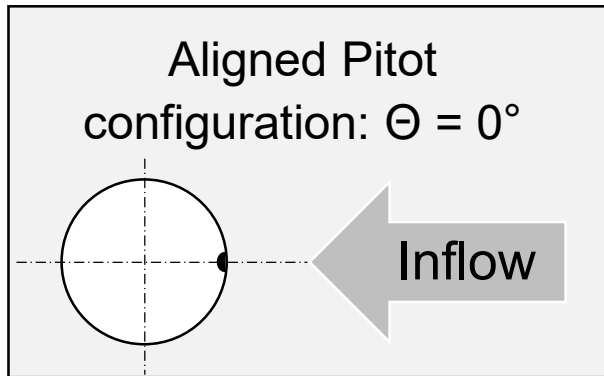
Wyler, J. S. (1975) Probe Blockage Effects in Free Jets and Closed Tunnels. ASME J. Eng. Gas Turbines Power, vol. 97(4), 509-514

# Experimental results

## Robustness of the cylinder Pitot probe against flow misalignment

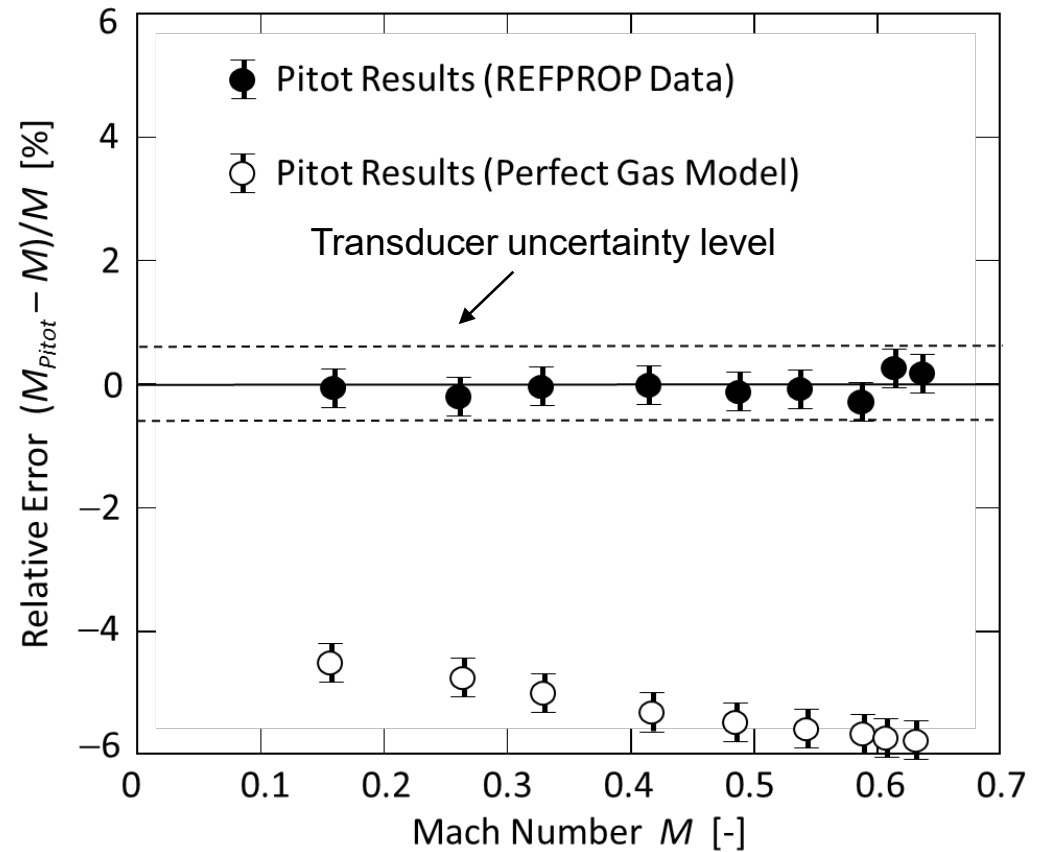


# Experimental results

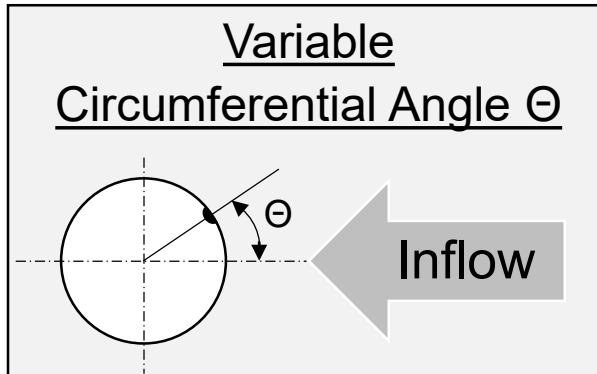


Reference

$$M = f(\dot{m}, p, p_0, T_0)_s$$



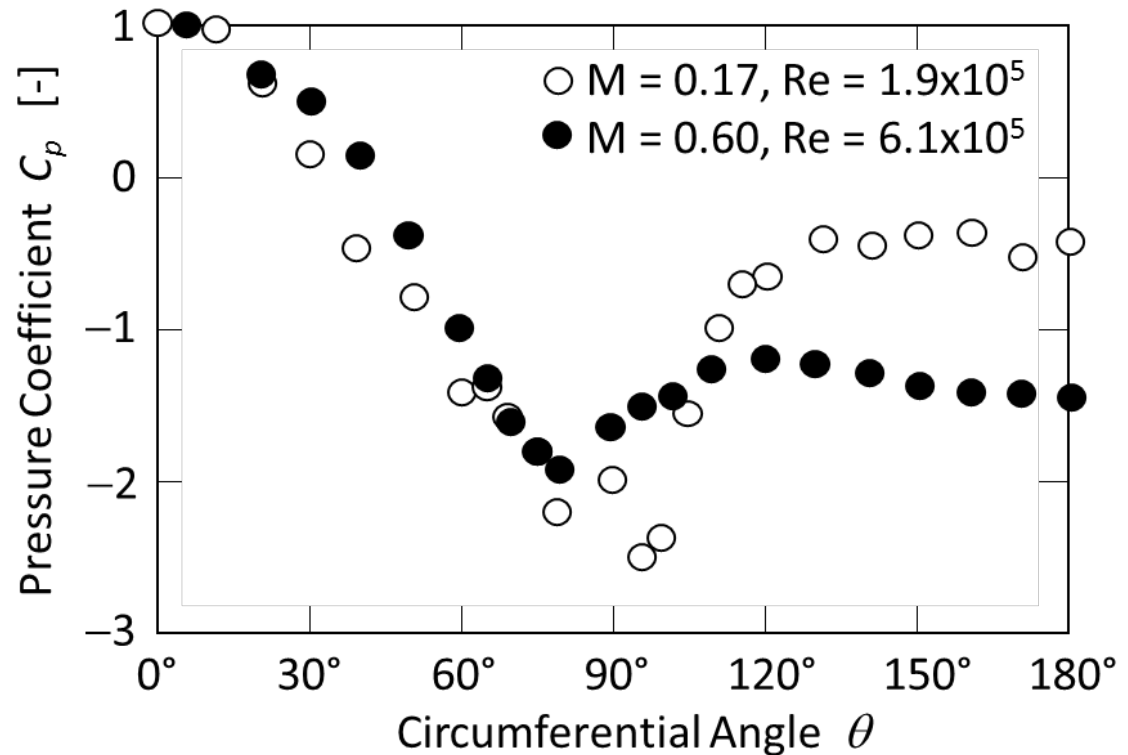
# Experimental results



Pressure Coefficient  $C_p$

$$C_p = \frac{p_{pitot} - p}{p_0 - p}$$

$p$ : freestream static pressure  
 $p_{pitot}$ : pitot pressure  
 $p_0$ : stagnation pressure

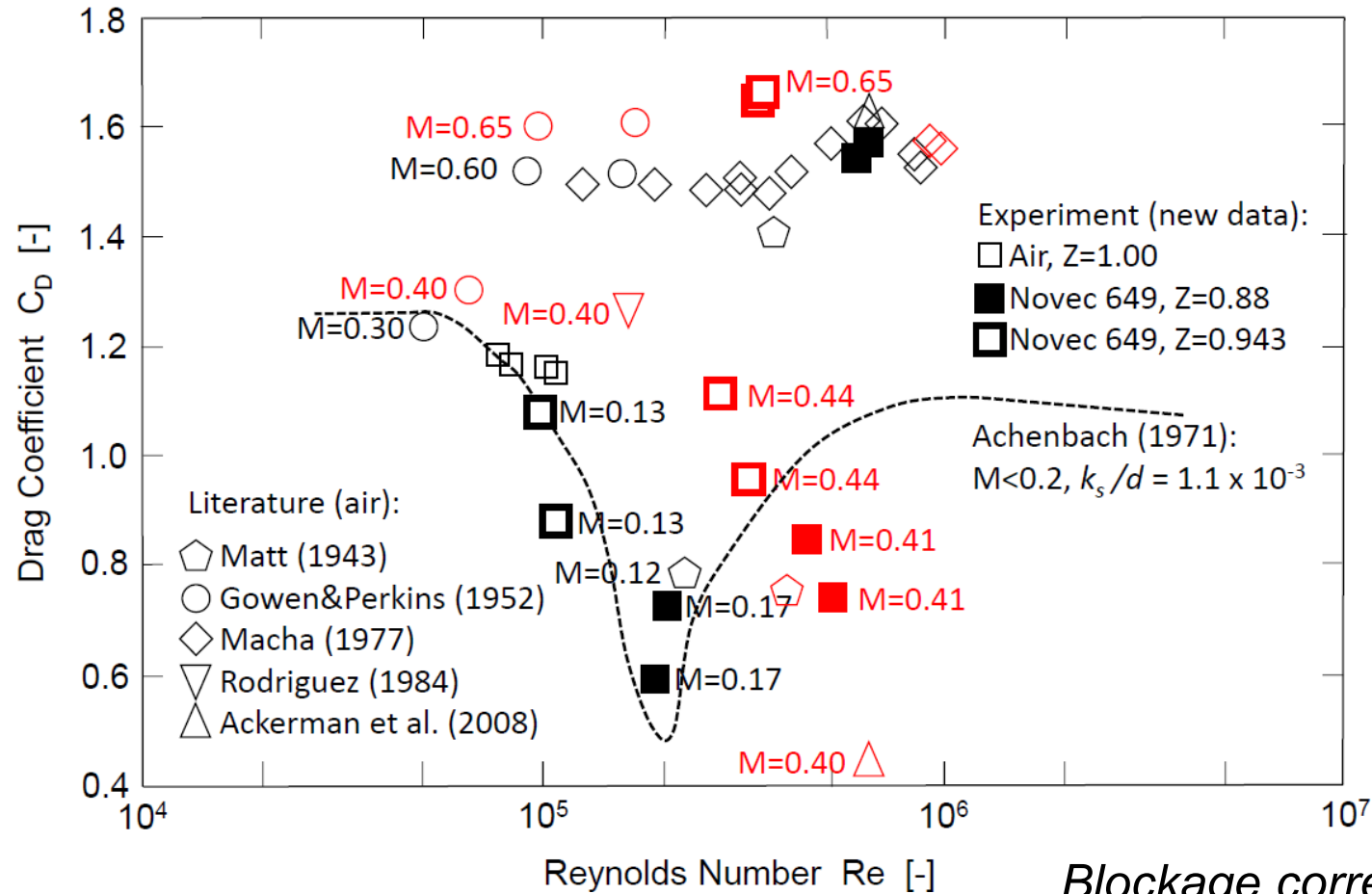


- ✓  $M=0.17$ : Agreement with literature data
- $M=0.6$ : Drag divergence → Transition



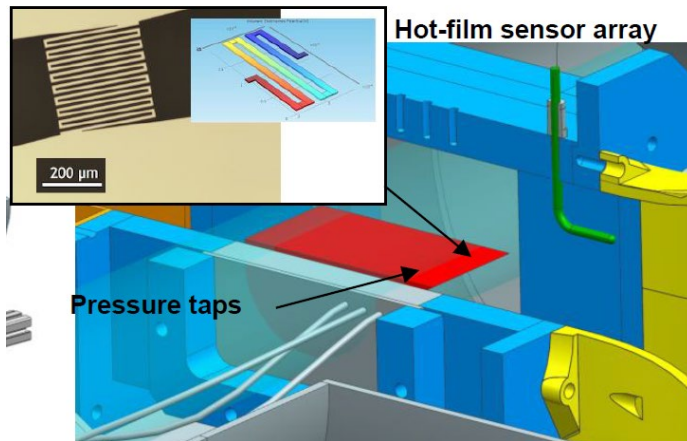
# Experimental results

Drag coefficient for Blockage Correction  
depends on Ma & Re (but not on Z)



# Conclusion & Outlook

- Mach number determination with Cylinder Pitot Probe was successful in high subsonic flow
- Probe Blockage effects have to be considered, especially for larger blockage factors (as usual in turbomachinery applications) to minimize uncertainties
- Drag divergence and locally supersonic flow phenomena were observed, which in turn affect probe blockage and should therefore be carefully considered
- Future research: see REGAL-ORC (2021-2024)



Thanks for your  
attention 😊

